

# Modeling of Velocity Distribution Among Microchannels with Triangle Manifolds

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DOI 10.1002/aic.11817

Published online June 23, 2009 in Wiley InterScience (www.interscience.wiley.com).

*A model of velocity distribution among microchannels with triangle manifolds is proposed. According to the flow behaviors analyzed by Fluent, the manifolds are divided into several approximate rectangular channels, and then an equivalent simplified resistance network model is developed to establish the relationships between the velocities and pressure drops in microchannels and approximate rectangular channels. The velocity distributions are calculated under two situations, respectively, considering and ignoring singular losses. The outcomes of the present study are compared with Fluent's simulated results to analyze the effects of singular losses on the velocity distributions. It indicates that the proposed model is suitable for the calculation of velocity distribution among microchannels with obtuse angled or right triangle manifolds under low Reynolds numbers. The premise of ignoring singular losses is that the frictional pressure drops are three times larger than the singular pressure drops in each flow loop. The manifold optimization results indicate that the velocity distribution among microchannels with right triangle manifolds is much more uniform than that of the corresponding one with obtuse angled manifolds. © 2009 American Institute of Chemical Engineers AIChE J, 55: 1969–1982, 2009*

**Keywords:** velocity distribution, laminated microreactor, microchannel array, triangle manifold

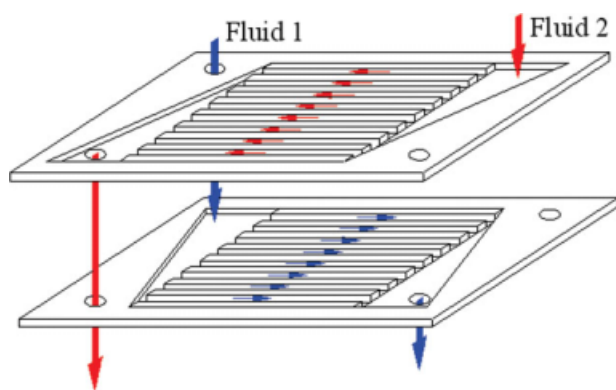
## Introduction

Microchannel reactors are chemical reaction units with microchannels fabricated by microfabrication technology on the solid surface. Incorporating microchannels offers higher heat and mass transfer rates, as well as higher surface-to-volume ratios, leading to improved process efficiency and smaller reaction volumes. Rapid development of microfabrication technology has enabled microchannel reactors to be

gradually applied in the fields of chemistry, chemical engineering, and biotechnology<sup>1</sup> within several years.

The two fundamentally different constructions of microreactors are monolithic construction and laminated-sheet structure. Microchannels or microstructures are incorporated into a single metal<sup>2–4</sup> or silicon block<sup>5,6</sup> by microfabrication methods to form monolithic construction, while multiple metal or silicon sheets, patterned with microchannel array, are first piled up and then welded together to yield a single laminated-sheet component.<sup>7,8</sup> Although the fabrication of monolithic construction is relatively simple, it can only serve a single reaction, and therefore, it is hard to integrate with complicated structures and multiple reactions.

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**Figure 1. The stacked microchannel sheets with mirror manifolds.**

[Color figure can be viewed in the online issue, which is available at [www.interscience.wiley.com](http://www.interscience.wiley.com).]

As laminated-sheet structure enables the coupling of endothermic and exothermic reactions in the same component, and moreover, the number of microchannel sheets can be increased or decreased on the demand of reactions. Laminated-sheet structure is considered as the best solution in terms of satisfying the complex requirements of microreactors.

The fluid velocity distribution among microchannels directly affects the performances of laminated microreactors. Relatively uniform velocity distribution is in favor of improving the heat and mass transfer efficiency and obtaining more even residence time, as well as improving the conversion rate and selectivity of process. Some preliminary theoretical analyses<sup>9,10</sup> on velocity or flow distribution among microchannels have been made. However, only sparse attempts have been reported in the literature concerning on the effects of singular losses on the velocity distribution among microchannels. For the purpose of achieving rapid calculation of fluid velocities, some researchers<sup>10,11</sup> neglected the singular losses among microchannels to obtain linear dependence of pressure drop on the velocity. However, it indicated that the effects of singular losses on total pressure could only be ignored under certain conditions.<sup>12</sup>

The objective of the present work is to develop a theoretical model of velocity distribution among microchannels with triangle manifolds and analyze the effects of singular losses on the velocity distribution. For this purpose, the manifolds are divided into several approximate rectangular channels based on the analysis of pressure distributions in the manifolds by Fluent, and then an equivalent simplified resistance network model is developed to establish the relationships between the velocities and pressure drops in microchannels and approximate rectangular channels. The velocity distributions are calculated under two situations, respectively, considering and ignoring singular losses. The calculation outcomes are compared with Fluent's simulated results to verify the validity of the proposed theoretical model and analyze the effects of singular losses on the velocity distributions. Moreover, the application scope of theoretical model is analyzed, and an optimization method of the manifold shape is proposed.

## Microchannel Array Model with Triangle Manifolds

### Model structure

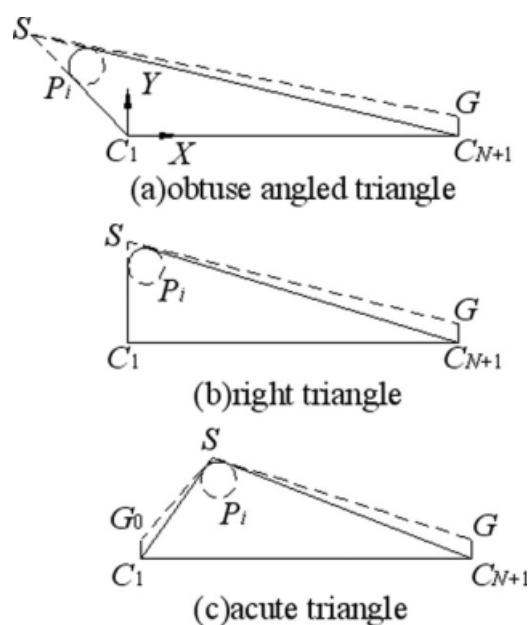
Multiple microchannel sheets with mirror manifolds are usually stacked together in a laminated microreactor for the coupling of endothermic and exothermic reactions, therefore, the manifold shape is generally designed to be triangle or trapezoid, leaving other side as the passage for another fluid, as shown in Figure 1.

According to the magnitude of  $\angle SC_1C_{N+1}$  of manifolds, the manifolds can be divided into three types, as shown in Figure 2. When the magnitude of  $\angle SC_1C_{N+1}$  is more than, equivalent to, or less than  $90^\circ$ , the manifolds here are called obtuse angled triangle, right triangle, or acute triangle manifold, respectively. Among these three types of manifolds, the similar polygon manifolds  $SGC_{N+1}C_1(G_0)$  are also included. The polygon manifold is turned into the triangle one when the lengths of  $GC_{N+1}$  and  $G_0C_1$  are equivalent to zero.

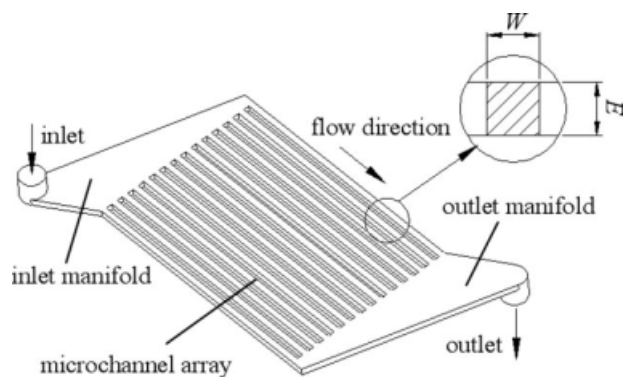
In the laminated structure as depicted in Figure 1, only a single microchannel sheet is selected to study the characteristics of velocity distribution among microchannels in this work. The velocity distributions in other laminated sheets can be obtained by the change of the flow behaviors, such as entrance velocity and kinetic viscosity. The studied model consists of an inlet, an outlet, parallel microchannels with rectangular cross-sections, and two triangle manifolds: one is inlet manifold and the other is outlet manifold, as shown in Figure 3. The direction of fluid is perpendicular to the inlet and outlet.

### Simulation analysis of pressure distributions in the manifolds

Because of irregular manifold shapes, it is difficult to study the flow pattern in the manifolds and it is necessary to simplify the manifolds. Take the obtuse angled triangle manifolds for example, as shown in Figure 2a. First, choose  $C_1$  as the origin of the coordinate system. Then select



**Figure 2. Three different types of manifolds.**



**Figure 3. Microchannel array model with triangle manifolds.**

$C_1C_{N+1}$  as the axis X and right as positive, while the vertical plumb perpendicular to  $C_1C_{N+1}$  as axis Y and up as positive. During the design of the manifold, the lengths of  $C_1C_{N+1}$  and  $GC_{N+1}$  are first determined, and then the position and magnitude of the inlet  $P_i$  are chosen, two tangent lines to the circle  $P_i$  through  $C_1$  and  $G$  are respectively made, which yields the final manifold shape. The lengths of manifold bottom  $C_1C_{N+1}$  and side  $GC_{N+1}$  are defined as  $L_m$  and  $H$ , respectively. And the microchannel number, length, width, depth, and interval are defined as  $N$ ,  $L_c$ ,  $W_c$ ,  $E$ , and  $W_s$ , respectively.

For the purpose of studying the flow pattern in the manifolds, a special case shown in Table 1 is selected here to analyze the pressure distributions in the inlet and outlet manifolds under different Reynolds numbers based on the inlet diameter. The liquid water (density as  $998.2 \text{ kg m}^{-3}$  and kinetic viscosity as  $1.003 \times 10^{-3} \text{ kg m}^{-1} \text{ s}^{-1}$ ) and gaseous water (density as  $0.5542 \text{ kg m}^{-3}$  and kinetic viscosity as  $1.34 \times 10^{-5} \text{ kg m}^{-1} \text{ s}^{-1}$ ) under 300 K are selected as the fluid, respectively. The Navier–Stokes equation with non-slip boundary condition and negligible gravity is used to evaluate the flow characteristics.<sup>11</sup> The model was meshed by Cooper scheme type, as depicted in Figure 4. The boundary conditions used are the velocity value in the Z direction of the inlet and the freedom outlet flow. The outlet pressure condition is set to zero and the solid boundaries are stationary. The flow pattern is assumed to be laminar in the microchannels. Segregated manner is selected as solver type.

**Table 1. Basic Structural Parameters of the Microchannel Model**

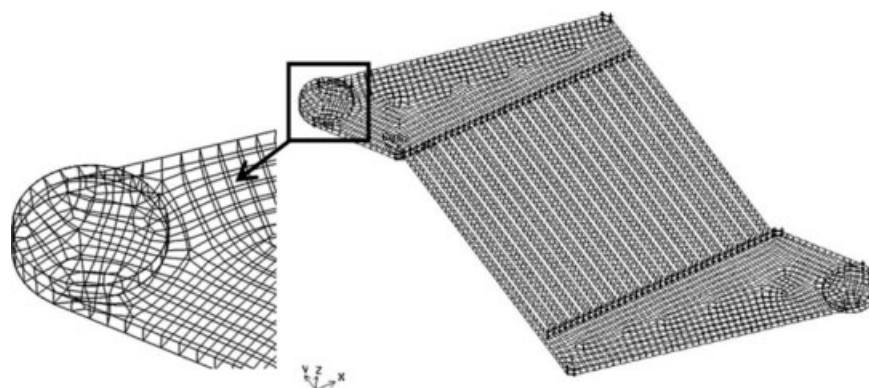
Microchannel Structural Parameters				
$N$	$L_c$	$W_c$	$E$	$W_s$
20	20 mm	$500 \mu\text{m}$	$500 \mu\text{m}$	$500 \mu\text{m}$
Manifold Structural Parameters				
$H$		coordinates of inlet $P_i(X_{pi}, Y_{pi})$		$R_{pi}$
2 mm		(-2,7)		2 mm

The simulated results of pressure distributions in the microchannel array model are shown in Figure 5. As presented in the figures, the laws of pressure distributions in the model accord to each other when the liquid water and gaseous water appear the same Reynolds numbers. When the Reynolds number is no more than 300, the pressure contours in the inlet and outlet manifold are approximately parallel to each other, and the directions of pressure contours are perpendicular to the bisector of  $\angle C_1SG$ . When the Reynolds number surpasses 300, the pressure contours in the inlet manifold become more unparallel with the increasing Reynolds numbers. Because of the coupling between non-linear flow singularities and velocity profiles under development, the pressure contours in the outlet manifold still maintain parallel to each other. The pressure distributions in the right triangle manifolds are similar to those of Figure 5.

#### Division method of manifolds

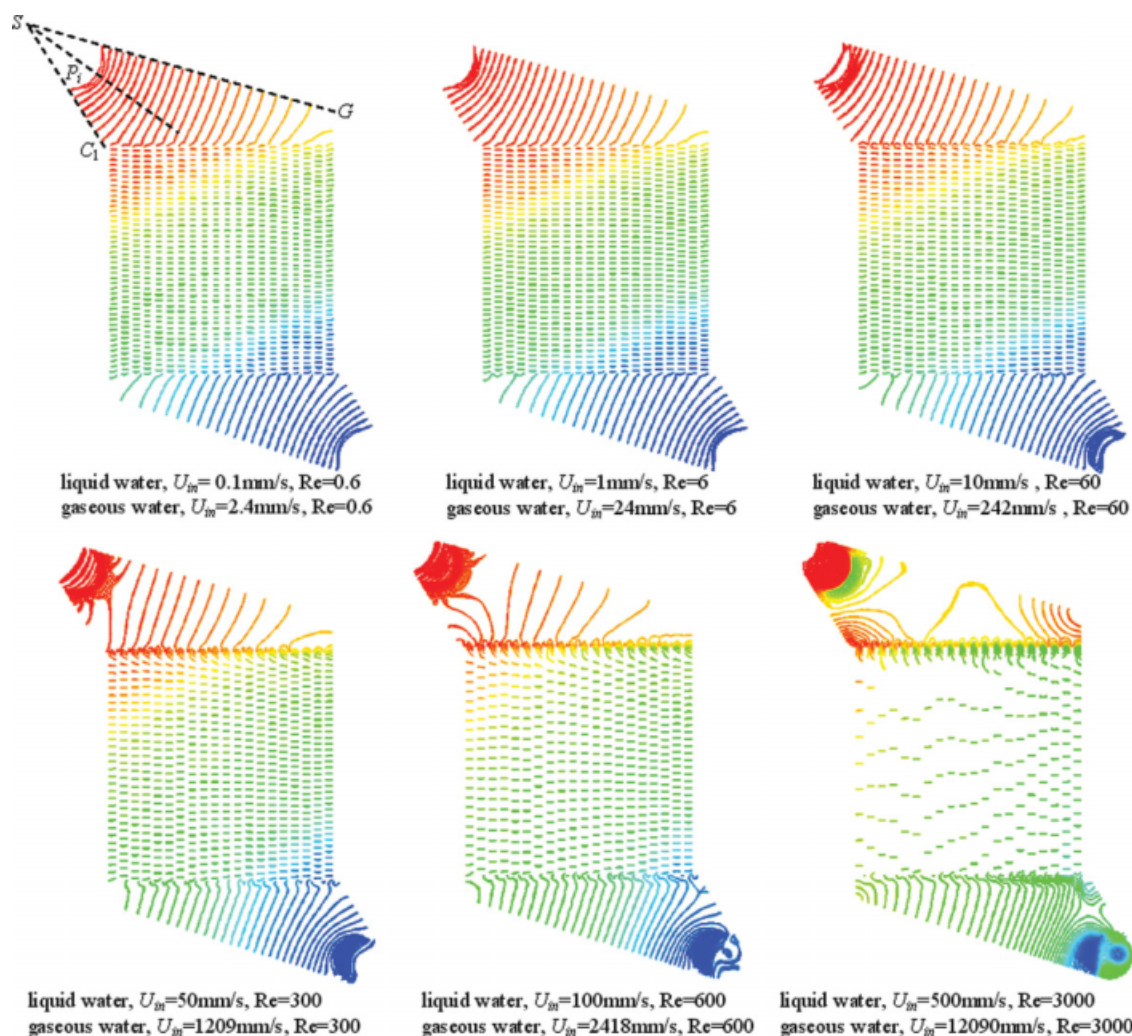
The direction of resultant forces affected on the fluid during the flowing process is perpendicular to the isobaric surface and points to the direction of the decreasing pressures, so the manifolds can be divided into multiple flow passages along the direction of the decreasing pressures, and the directions of flow passages are perpendicular to the pressure contours. Since the pressure contours are perpendicular to the bisector of  $\angle C_1SG$  under low Reynolds numbers, the divided flow passages are also perpendicular to the bisector of  $\angle C_1SG$ .

Take the inlet manifold  $GT_1T_2C_1C_{N+1}$  for example to introduce the division method of manifolds, as shown in Figure 6. The division method of the outlet manifold is similar



**Figure 4. The model meshed by Cooper scheme type.**





**Figure 5. Pressure distributions in the microchannel array model under different Reynolds numbers ( $U_{in}$  refers to the entrance velocity).**

[Color figure can be viewed in the online issue, which is available at [www.interscience.wiley.com](http://www.interscience.wiley.com).]

to that of the inlet manifold. First, extend the segment  $GT_1$  and  $C_1T_2$  to  $S$ , and then draw the bisector  $SP_1$  of  $\angle GSC_1$ . Through  $C_1$ ,  $C_1F_1$  is drawn perpendicular to  $SP_1$  and meets the segment  $SG$  in  $F_1$ , which divides the inlet manifold into two parts: one is the polygon  $T_1T_2C_1F_1$  called transition chamber here, and the other is the polygon  $C_1C_{N+1}GF_1$  called connected chamber. Then the connected chamber  $C_1C_{N+1}GF_1$  is further divided into multiple intervals. Draw  $C_iF_i$  ( $i = 2, 3, \dots, N$ ) perpendicular to  $SP_1$  through  $C_i$  ( $i = 2, 3, \dots, N$ ), and intersect to the segment  $GS$  at  $F_i$  ( $i = 2, 3, \dots, N$ ), respectively. Through  $F_i$  ( $i = 1, 2, \dots, N-1$ ), draw vertical plumb perpendicular to the segment  $C_{i+1}F_{i+1}$  ( $i = 1, 2, \dots, N-1$ ). Therefore, the connected chamber  $C_1C_{N+1}GF_1$  is divided into  $N$  intervals, which forms  $N$  approximate rectangular channels  $C_iC_{i+1}F_{i+1}F_i$  ( $i = 1, 2, \dots, N$ ), and the  $N$ th channel  $C_NC_{N+1}F_N$  also seems as an approximate rectangular channel here even though it is a triangle channel.

After the microchannel array model being simplified, by the calculation methods presented in Appendix A, the widths

and lengths of approximate rectangular channels can be obtained as follows

$$L_{C_iF_i} = \sqrt{(X_{C_i} - X_{F_i})^2 + Y_{F_i}^2} \quad (i = 1, 2, \dots, N) \quad (1)$$

$$L_{in} = L_m / N \cdot \sin(\arctan k_{C_iF_i}) \quad (2)$$

where

$$\begin{cases} X_{F_i} = \frac{k_{SG} \cdot L_m - k_{C_iF_i} \cdot (i-1) \frac{L_m}{N} - H_{in}}{k_{SG} - k_{C_iF_i}} \\ Y_{F_i} = \frac{-k_{SG} k_{C_iF_i} (i-1) \frac{L_m}{N} + k_{SG} k_{C_iF_i} L_m - k_{C_iF_i} H_{in}}{k_{SG} - k_{C_iF_i}} \end{cases} \quad (3)$$

If  $X_{F_i}$  (obtained by Eq. 3)  $> L_m$ , then

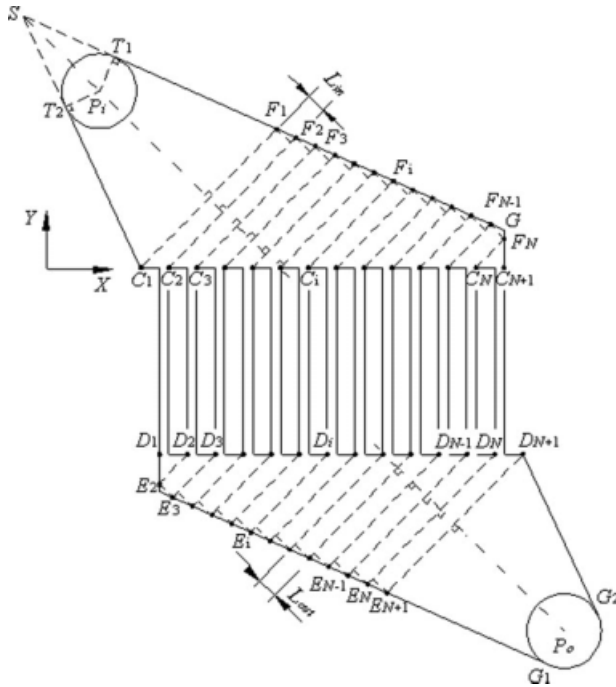


Figure 6. Division method of manifolds.

$$\begin{cases} X_{Fi} = L_m \\ Y_{Fi} = k_{Ci} (L_m - X_{Ci}) \end{cases} \quad (4)$$

$$X_{Ci} = (i-1) \frac{L_m}{N} \quad (5)$$

Here  $k$  represents the slope coefficient of each line segment.

### Analysis of Flow Resistance Losses

Two categories of energy losses result from flow resistances during the flowing process: one is frictional losses caused by the wall shear in the pipe, and the other is singular losses caused by a change in the magnitude or the direction of the velocity vectors at the bends, enlargements, and contractions.

#### Frictional loss and pressure drop

In practice, the flow velocity in the microchannel is generally less than  $10 \text{ m s}^{-1}$  and the equivalent diameter of microchannel is no more than  $500 \mu\text{m}$  so that the Reynolds number is less than 1000. Therefore, the flow pattern in microchannels is considered as a laminar flow.

According to Hagen-Poiseuille Equation, for a laminar flow in rectangular channels, the pressure drop  $\Delta P$  due to frictional losses is defined as

$$\Delta P = \frac{32\mu L \lambda_{NC}}{D_H^2} U = R_f U \quad (6)$$

where  $L$  is the channel length and  $D_H$  is hydraulic nominal diameter which is defined as below

$$D_H = \frac{2WE}{W+E} \quad (7)$$

$\lambda_{NC}$  is the non-circular coefficient,<sup>9,10</sup> of which magnitude depends on the ratio of channel depth  $E$  and width  $W$  as follows

$$\lambda_{NC} = \frac{\frac{3}{2}}{[1 - 0.351 \min(\frac{E}{W}, \frac{W}{E})]^2 [1 + \min(\frac{E}{W}, \frac{W}{E})]^2} \quad (8)$$

Substituting Eqs. 7 and 8 into Eq. 6, the following equation can be obtained.

$$R_f = \frac{12\mu L}{\min^2(W, E) [1 - 0.351 \min(\frac{E}{W}, \frac{W}{E})]^2} \quad (9)$$

Here  $R_f$  is called frictional resistance.

#### Singular loss and pressure drop

To analyze the catalogues of singular losses in the microchannel array model, the  $j$ th flow loop is selected from the model, as shown in Figure 7. The  $j$ th flow loop contains the  $j$ th approximate rectangular channel of the inlet manifold, the  $(j-1)$ th approximate rectangular channel of the outlet manifold, the  $(j-1)$ th and  $j$ th microchannel. Fluid in the flow loop flows in two different directions: the fluid from the  $j$ th approximate rectangular channel of the inlet manifold is split into two parts, one flow in the  $j$ th microchannel, and the other flow in the  $(j+1)$ th approximate rectangular channel of the inlet manifold, while the fluid from the  $(j-1)$ th microchannel and the  $(j-2)$ th approximate rectangular channel of the outlet manifold are combined into the  $(j-1)$ th approximate rectangular channel of the outlet manifold.

According to the analysis of the flow process of fluid in the  $j$ th flow loop, two kinds of singular losses<sup>13</sup> exist in the microchannel array model:

(1) As the fluid flows from the approximate rectangular channels of the inlet manifold to microchannels, there are three singular losses: diffuser loss  $h_{jd}$  caused by fluid

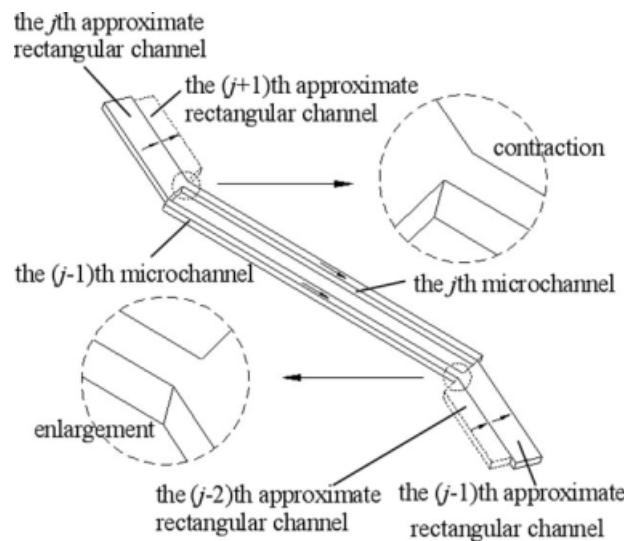


Figure 7. The  $j$ th flow loop.

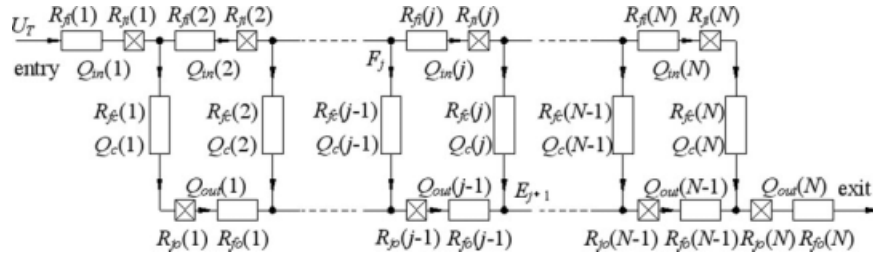


Figure 8. Simplified resistance network model.

splitting, contraction loss  $h_{jc}$  caused by the sudden contracting section and turning loss  $h_{jti}$ .

(2) When the fluid flows from microchannels to the approximate rectangular channels of the outlet manifold, the singular losses are composed of mixing loss  $h_{jm}$  caused by combination of two fluids with different velocities, enlargement loss  $h_{je}$  caused by the sudden increasing section and turning loss  $h_{jto}$ .

$\Delta P_{ji}(j)$  refers to the pressure drop caused by singular losses as the fluid flows from the  $j$ th approximate rectangular channels of the inlet manifold to the  $j$ th microchannels while  $\Delta P_{jo}(j-1)$  refers to the pressure drop caused as the fluid flows from the  $(j-1)$ th microchannels to the  $(j-1)$ th approximate rectangular channels of the outlet manifold. They are defined, respectively, as follows.

$$\Delta P_{ji}(j) = (\zeta_{jd} + \zeta_{jc} + \zeta_{jti}) \frac{\rho}{2} U_{in}^2(j) = R_{ji}(j) U_{in}^2(j) \quad (j = 1, 2, \dots, N) \quad (10)$$

$$\Delta P_{jo}(j-1) = (\zeta_{jm} + \zeta_{je} + \zeta_{jto}) \frac{\rho}{2} U_c^2(j-1) = R_{jo}(j-1) U_c^2(j-1) \quad (j = 2, 3, \dots, N+1) \quad (11)$$

where,  $\zeta_{jd}$ ,  $\zeta_{jc}$ ,  $\zeta_{jti}$ ,  $\zeta_{jm}$ ,  $\zeta_{je}$ , and  $\zeta_{jto}$  are singular resistance coefficients corresponding to singular losses  $h_{jd}$ ,  $h_{jc}$ ,  $h_{jti}$ ,  $h_{jm}$ ,  $h_{je}$ , and  $h_{jto}$ .  $R_{ji}$  and  $R_{jo}$  are called the singular resistances of entry end and exit end of the flow loop, respectively.

### Simplified resistance network model

To distinctly analyze the relationships between the velocities and pressure drops in microchannels and approximate rectangular channels, the microchannel array model is equivalent to a simplified resistance network model, as depicted in Figure 8. In this model, the frictional resistances in the microchannels, the approximate rectangular channels of the inlet and outlet manifold are represented as electrical resistances  $R_{fc}(j)$ ,  $R_{fi}(j)$ , and  $R_{fo}(j)$  ( $j = 1, 2, \dots, N$ ), respectively. The singular resistances of entry end and exit end of the flow loop are equivalent to  $R_{ji}(j)$  and  $R_{jo}(j)$  ( $j = 1, 2, \dots, N$ ).

Transition chamber  $T_1T_2C_1F_1$  of the inlet manifold and  $D_{N+1}E_{N+1}G_1G_2$  of the outlet manifold are not considered in the simplified resistance network model, because they show no influences on the flow distribution among microchannels. To guarantee that the two models could be consistent to each other, according to the continuity equation of incompressible fluid, the entrance velocity  $U_T$  of the simplified resistance net-

work model and the entrance velocity  $U_{in}$  of microchannel array model should satisfy the following relation.

$$U_{in} A_{pi} = U_T A_{T_1T_2C_1F_1} \quad (12)$$

where  $A_{pi}$  and  $A_{T_1T_2C_1F_1}$  are the sectional areas of inlet  $P_i$  and the exit of transition chamber  $T_1T_2C_1F_1$ , respectively.

In the simplified resistance network model, the pressure drops of two ends (such as point  $F_j$  to  $E_{j+1}$  in Figure 8) in the diagonal line of each flow loop via two different flow channels are equal, therefore  $N-1$  equations can be inferred as follows

$$\Delta P_{fi}(j) + \Delta P_{ji}(j) + \Delta P_{fc}(j) = \Delta P_{fc}(j-1) + \Delta P_{jo}(j-1) + \Delta P_{fo}(j-1) \quad (j = 2, 3, \dots, N) \quad (13)$$

According to the continuity equation of incompressible fluid, as shown in the simplified resistance network model, the relations of flow rates in the approximate rectangular channels of the inlet manifold and microchannels are defined as follows

$$Q_{in}(j) = Q_c(j) + Q_{in}(j+1) \quad (j = 1, 2, \dots, N-1) \quad (14)$$

$$Q_{in}(1) = Q_T \quad (15)$$

$$Q_{in}(N) = Q_c(N) \quad (16)$$

where  $Q_T$  is the total feed flow rate.

The relations of flow rates in the approximate rectangular channels of the outlet manifold and microchannels are defined as follows

$$Q_{out}(j) = Q_c(j) + Q_{out}(j-1) \quad (j = 2, 3, \dots, N) \quad (17)$$

$$Q_{out}(1) = Q_c(1) \quad (18)$$

The relationships between the channel velocity  $U(j)$  and flow rate  $Q(j)$  can be defined as follows

$$U(j) = \frac{Q(j)}{WE} \quad (j = 1, 2, \dots, N) \quad (19)$$

Substituting Eq. 19 into Eqs. 14–18, then  $2N + 1$  equations are established about  $U_{in}(j)$ ,  $U_{out}(j)$ , and  $U_c(j)$ . Together with Eq. 13, there are  $3N$  independent equations. Since there



are  $3N$  unknown variables about  $U_{in}(j)$ ,  $U_{out}(j)$ , and  $U_c(j)$  in the simplified resistance network model, they can be worked out by the combination of Eqs. 13–19.

### Simplified calculation method of singular resistance coefficients

If all the structural parameters of the microchannel model are given, frictional resistance  $R_f$  can be calculated by Eq. 9, but the singular resistances cannot be solved according to Eqs. 10 and 11 unless the values of  $\zeta_{jd}$ ,  $\zeta_{jc}$ ,  $\zeta_{ji}$ ,  $\zeta_{jm}$ ,  $\zeta_{je}$ , and  $\zeta_{jto}$  are obtained. Equation 13 is established on the basis of  $N-1$  flow loops and mainly relate to singular losses  $h_{jc}$ ,  $h_{ji}$ ,  $h_{je}$ , and  $h_{jto}$  in a flow loop, therefore the diffuser loss  $h_{jd}$  and mixing loss  $h_{jm}$  are ignored here.

The values of singular resistance coefficients in multiple microchannels are difficult even impossible to be exactly measured by experiments. Nowadays, the existed empirical equations of the singular resistance coefficients are just suitable for a single microchannel, there are no empirical equations for the complicated microchannel array proposed in this work. Even if the empirical equations of the complicated microchannel array were known, the calculations of singular resistance coefficients by empirical equations need to determine the velocities in the approximate rectangular channels and microchannels in advance. However, these values are unknown and require to be solved. Therefore, a simplified calculation method of singular resistance coefficients is proposed in this work.

Here,  $\zeta_{jc}$  and  $\zeta_{je}$  are calculated by the empirical equations of contraction loss and enlargement loss as follows, respectively.

$$\zeta_{jc}(j) = 0.5[1 - A_c(j)/A_{in}(j)] \quad (j = 2, 3, \dots, N) \quad (20)$$

$$\zeta_{je}(j) = [1 - A_c(j-1)/A_{out}(j-1)]^2 \quad (j = 2, 3, \dots, N) \quad (21)$$

where  $A_c(j)$  represents the sectional area of the  $j$ th microchannel,  $A_{in}(j)$  and  $A_{out}(j-1)$  are the sectional areas of the  $j$ th approximate rectangular channel in the inlet manifold and the  $(j-1)$ th approximate rectangular channel in the outlet manifold, respectively.

Nowadays, few refers are provided for the calculations of  $\zeta_{ji}$  and  $\zeta_{jto}$  on the microscale. In this work,  $\zeta_{ji}$  and  $\zeta_{jto}$  are calculated by the empirical formula of the turning loss coefficients in the acute angle corner pipe with rectangular cross-section on the large scale<sup>14</sup> as follows, and then the validity are verified by the comparison of the calculation results of velocity distributions to the simulation ones.

$$\zeta_T = C_T A_T \zeta_M \quad (22)$$

where  $C_T$  is a function of the depth-width ratio  $E/W$  of channels, and  $A_T$  and  $\zeta_M$  are the influence coefficient of turn angle  $\delta$  of channels. The values of these three coefficients can be referred in Appendix B.

## Results and Discussion

### Velocity distribution profiles

Under different Reynolds numbers, the comparisons of the velocity distribution among microchannels simulated by

Fluent and calculated by two methods of the theoretical model are shown in Figure 9. Results indicate that the two calculation methods accord with the Fluent's simulation under low Reynolds numbers, such as 0.6 and 6, which shows that the singular losses take little effects on the velocity distribution among microchannels. The velocity distribution appears symmetrical. Commenge et al.<sup>9</sup> considered that the centrosymmetry of the microchannel model was the cause of a symmetrical velocity distribution.

With larger Reynolds numbers, such as 60 and 300, the calculations considering singular losses also maintain consistency with the simulation one. The velocity distributions begin to be asymmetrical, and the values of velocities in the microchannels far away from the inlet are larger than that in the microchannels near the inlet. This trend becomes more obvious with the increasing Reynolds numbers. However, the velocity distributions obtained by the calculation which ignores singular losses still maintain symmetrical and begin to deviate from Fluent's simulation. The deviation becomes much larger with the increasing Reynolds numbers. As a result, the singular losses play an important role under large Reynolds numbers.

With further increasing Reynolds numbers, such as 600 and 3000, both calculation results, especially the one ignoring the singular losses, significantly deviate from the simulation one. This is because the pressure contours in the manifold become unparallel under large Reynolds numbers and hence results in unsuitable application of the theoretical model proposed based on parallel distributions of pressure contours.

From the comparison results of the velocity distributions among microchannels calculated by the theoretical model and simulated by Fluent, it can be concluded that the calculation of  $\zeta_{js}$  and  $\zeta_{jd}$  by the formula of the turning loss coefficients in the acute angle corner pipe with rectangular cross-section on the large scale are feasible under low Reynolds numbers.

### Analysis of the effects of singular losses

The effects of singular losses under low Reynolds numbers are analyzed here by way of a function  $\beta(j)$  of pressure drop proportion in each flow loop, which are defined below

$$\beta_i(j) = \frac{\Delta P_{fi}(j)}{\Delta P_{ji}(j)} \quad (j = 2, 3, \dots, N) \quad (23)$$

$$\beta_o(j-1) = \frac{\Delta P_{fo}(j-1)}{\Delta P_{jo}(j-1)} \quad (j = 2, 3, \dots, N) \quad (24)$$

where  $\beta_i(j)$  and  $\beta_o(j-1)$  represent the ratios of the pressure drop caused by frictional losses and singular losses in the inlet manifold and outlet manifold of the  $j$ th flow loop, respectively.

Table 2 shows the calculation results of pressure drop proportions with four different low Reynolds numbers. For all the flow loops, when the Reynolds number is very low, such as 0.6, the magnitudes of pressure drops in the inlet manifold and outlet manifold caused by frictional losses are 274–1748 and 65–253 times, respectively, as large as that caused

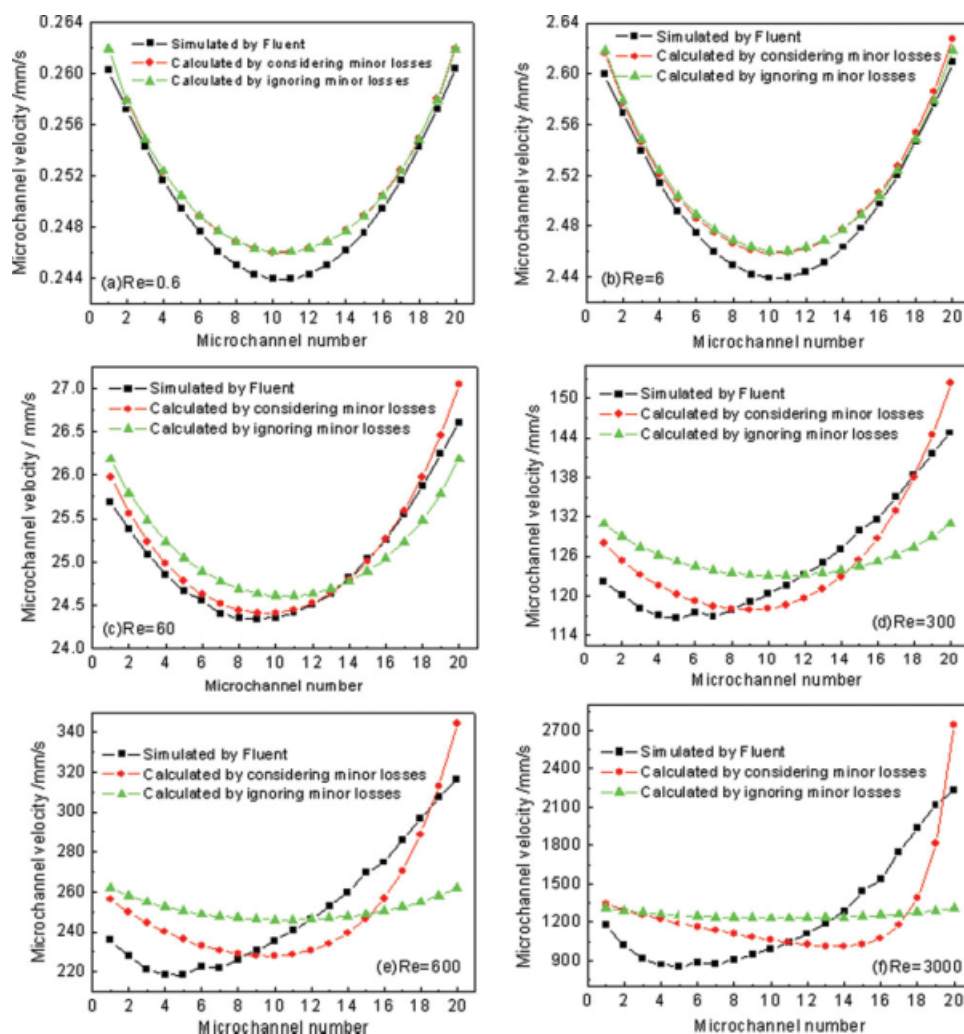


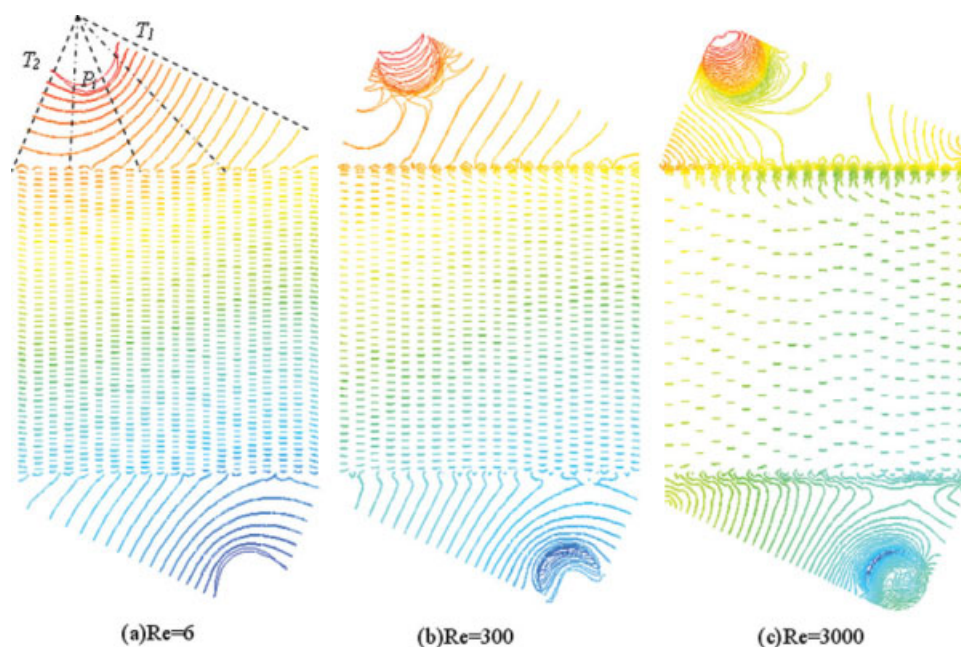
Figure 9. The comparisons of velocity distributions among microchannels calculated by two methods and simulated by Fluent under different Reynolds numbers.

[Color figure can be viewed in the online issue, which is available at [www.interscience.wiley.com](http://www.interscience.wiley.com).]

Table 2. The Calculation of Pressure Drop Proportion in Each flow Loop

$j$	$Re = 0.6$		$Re = 6$		$Re = 60$		$Re = 300$	
	$\beta_i(j)$	$\beta_o(j-1)$	$\beta_i(j)$	$\beta_o(j-1)$	$\beta_i(j)$	$\beta_o(j-1)$	$\beta_i(j)$	$\beta_o(j-1)$
2	274.86	65.45	27.49	6.55	2.75	0.66	0.55	0.13
3	279.95	111.09	27.99	11.12	2.80	1.12	0.56	0.23
4	285.46	144.64	28.54	14.48	2.85	1.46	0.57	0.30
5	291.52	170.25	29.14	17.04	2.91	1.72	0.58	0.36
6	298.16	190.35	29.81	19.05	2.97	1.93	0.59	0.40
7	305.74	206.32	30.56	20.65	3.04	2.09	0.60	0.43
8	314.35	219.24	31.41	21.95	3.13	2.22	0.62	0.46
9	324.19	229.71	32.40	22.99	3.22	2.32	0.63	0.48
10	335.85	238.22	33.56	23.84	3.33	2.40	0.65	0.50
11	349.79	245.02	34.94	24.52	3.46	2.47	0.67	0.51
12	366.76	250.44	36.63	25.05	3.63	2.51	0.70	0.52
13	387.87	254.58	38.74	25.46	3.83	2.55	0.74	0.52
14	415.07	257.61	41.45	25.75	4.09	2.56	0.78	0.52
15	451.48	259.54	45.07	25.92	4.44	2.57	0.84	0.51
16	502.42	260.38	50.16	26.00	4.93	2.56	0.92	0.50
17	579.61	260.24	57.82	25.96	5.66	2.54	1.05	0.48
18	708.22	259.02	70.63	25.82	6.90	2.51	1.26	0.46
19	966.06	256.60	96.37	25.56	9.39	2.46	1.69	0.43
20	1747.97	252.93	173.91	25.17	16.89	2.40	3.00	0.40





**Figure 10. Pressure distributions in the acute angled triangle manifolds under different Reynolds numbers.**

[Color figure can be viewed in the online issue, which is available at [www.interscience.wiley.com](http://www.interscience.wiley.com).]

by singular losses. When the Reynolds number is equal to 6, the ratios turn to be 27.5–174 times and 6.6–25 times, respectively. Evidently, the pressure drops caused by singular losses are very small under low Reynolds numbers, leading to little effects on the calculations of theoretical model, as shown in Figures 9a, b.

However, with the increasing Reynolds numbers, such as 60, the ratios of two kinds of pressure drops are changed to 2.7–16.9 times in the inlet manifold and 0.6–2.4 times in the outlet manifold, which indicates that the ratios of pressure drops decrease. If singular losses are ignored at the time, the calculation results will deviate from the original model, as shown in Figure 9c. And the deviation will be much larger with the increasing Reynolds numbers. When the Reynolds number is equal to 300, the ratios of two kinds of pressure drops are 0.5–3 times in the inlet manifold and 0.13–0.4 times in the outlet manifold, respectively. And the pressure drop caused by singular losses is nearly the same as the one caused by frictional losses, even more than the latter. Ignoring singular losses during the calculations will result in incorrect velocity distributions, as shown in Figure 9d.

Therefore, the governing factor for ignoring the singular losses under low Reynolds numbers is that the pressure drops caused by singular losses must be considerably small. On the basis of above analysis, so-called “considerably small” here means that the pressure drops caused by frictional losses should be three times larger than that caused by singular losses in each flow loop.

#### **The application scope of the theoretical model**

Figure 10 shows the distributions of pressure contours in the acute triangle manifolds under different Reynolds numbers. For the low Reynolds numbers, such as 6, the pressure

contours around the inlet appear to be curving. The distribution of pressure contours becomes somewhat disordered when  $Re = 300$ . And it becomes more disordered with the increasing Reynolds numbers. Obviously, the distributions of pressure contours in the acute triangle manifold are quite different from the one in the obtuse angled triangle manifolds. The theoretical model is established on the premise of parallel distributions of pressure contours, therefore the proposed theoretical model is not suitable for the microchannel array model with acute triangle manifolds. Otherwise, the obtained velocity distribution will deviate from the realities. Figure 11 shows the comparison of velocity distribution simulated by Fluent and the calculation by the theoretical model considering singular losses for the microchannel array with acute triangle manifolds. Here,  $U_m$  refers to the average velocity of microchannels as defined below.

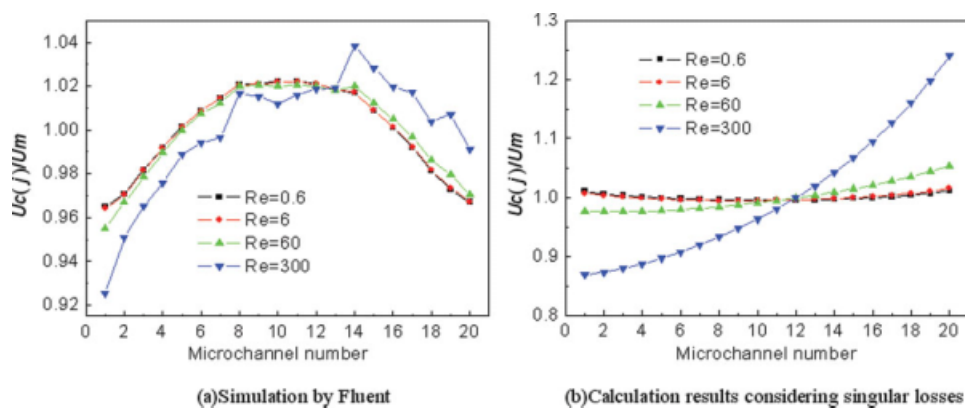
$$U_m = \frac{1}{N} \sum_{j=1}^N U_c(j) \quad (j = 1, 2, \dots, N) \quad (25)$$

#### **Optimization of the manifold shape**

For qualitatively estimating the quality of velocity distributions for different microchannel models, a parameter  $\sigma_U\%$  is defined as below. Smaller  $\sigma_U\%$  suggests more uniform flow distributions.

$$\sigma_U\% = 100 \sqrt{\frac{1}{N} \sum_{j=1}^N \left( \frac{U_c(j)}{U_m} - 1 \right)^2} \quad (26)$$

When the microchannel structural parameters are given, the velocity distributions among microchannels are mainly dominated by the manifold shape. From the point of



**Figure 11. Velocity distributions among microchannels with acute triangle manifolds under different Reynolds numbers.**

[Color figure can be viewed in the online issue, which is available at [www.interscience.wiley.com](http://www.interscience.wiley.com).]

manifold design, the manifold shape are determined by the position ( $X_{pi}$ ,  $Y_{pi}$ ) and the radius  $R_{pi}$  of the inlet (or outlet) as well as the side length  $H$  of the manifold. Therefore, the magnitudes of  $X_{pi}$ ,  $Y_{pi}$ ,  $R_{pi}$ , and  $H$  play an important role in the velocity distributions among microchannels. However, their values are usually limited within certain ranges due to the sheet dimension or fabrication constrains. Assume that their ranges are defined as follows, respectively.

$$\min X_{pi} \leq X_{pi} \leq \max X_{pi} \quad (27)$$

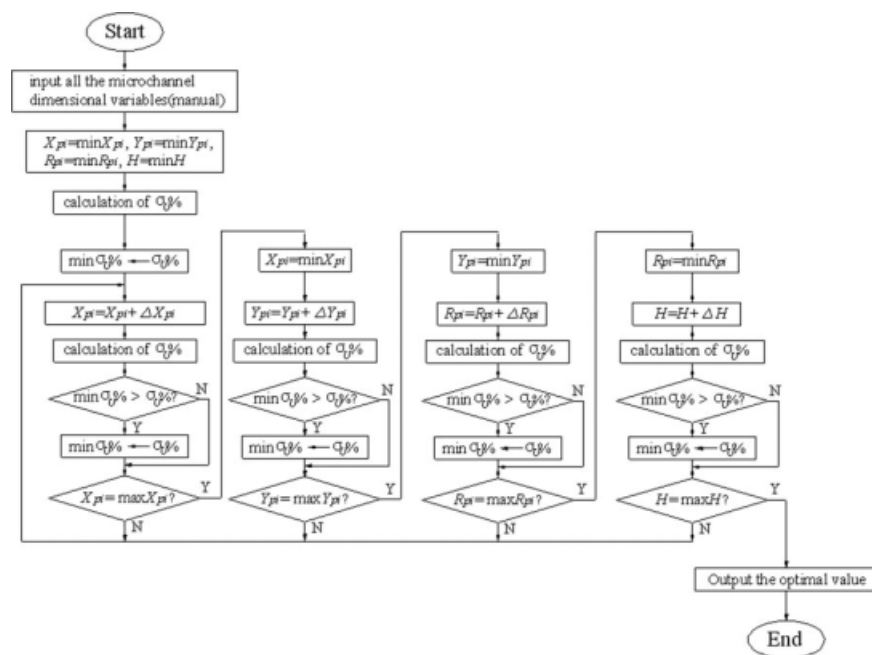
$$\min Y_{pi} \leq Y_{pi} \leq \max Y_{pi} \quad (28)$$

$$\min R_{pi} \leq R_{pi} \leq \max R_{pi} \quad (29)$$

$$\min H \leq H \leq \max H \quad (30)$$

The manifold shape is changed with the variety of  $X_{pi}$ ,  $Y_{pi}$ ,  $R_{pi}$ , and  $H$  within their limits. During the optimization of the manifold shape, all the  $\sigma_U\%$  values of different manifold shapes are worked out, and the shape corresponding to the minimum value of  $\sigma_U\%$  is the best one. The flowchart of the optimization procedure is shown in Figure 12.

However, numerous manifold shapes can be obtained with the simultaneous varieties of the four manifold structural parameters. It requires nevertheless lengthy time to reach the optimal shape. Therefore, only one of the structural parameters is usually chosen to optimize in practice while other parameters maintain constant. Here, the microchannel array model in Table 1 is also selected as example. Assume that the entrance velocity is  $1 \text{ mm s}^{-1}$  and the values of  $Y_{pi}$ ,  $R_{pi}$ , and  $H$  are invariant, the value of  $X_{pi}$  is increased stepwise from  $-4$  to  $2$ , with the increment of  $0.1$ . Figure 13 shows



**Figure 12. The flowchart of the optimization procedure.**

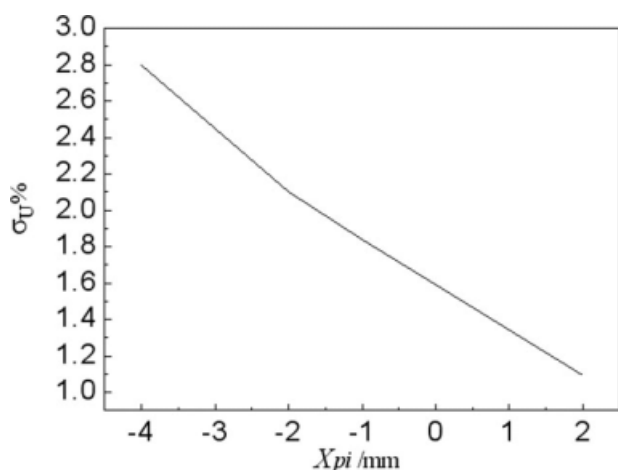


Figure 13. The relation between the  $\sigma_U\%$  and  $X_{pi}$ .

how the different  $\sigma_U\%$  values vary with different  $X_{pi}$ . It indicates that the velocity distributions among microchannels get more uniform with larger  $X_{pi}$ , and the velocity distributions among the microchannels with the right triangle manifolds ( $X_{pi} = 2$ ) is much more uniform than that of the corresponding one with obtuse angled manifolds ( $X_{pi} < 2$ ).

## Conclusions

The two calculation methods, considering and ignoring singular losses accord with the Fluent's simulation under low Reynolds numbers, but with larger Reynolds numbers, the calculation ignoring singular losses begins to deviate from the simulation by Fluent and the one considering singular losses also maintain consistence with the simulation. The singular losses can be ignored during the calculation as the frictional pressure drops are three times larger than the singular pressure drops in each flow loop.

With large Reynolds numbers, the calculation results seriously deviate from the simulation due to unparallel distributions of the pressure contours in the manifold. The pressure contours in acute triangle manifolds appear to be curving around the inlet under low Reynolds numbers. The theoretical model, which is established on the premise of parallel distributions of pressure contours, is only suitable for the calculation of velocity distributions among microchannels with obtuse angled or right triangle manifolds under low Reynolds numbers.

The manifold optimization results indicate that the velocity distribution among microchannels with right triangle manifolds is much more uniform than that of the corresponding one with obtuse angled manifolds when they have the same microchannel structural parameters and  $Y_{pi}$ ,  $R_{pi}$ , and  $H$ .

## Acknowledgments

This research was supported by the National Nature Science Foundation of China, Project No. 50805052, the Natural Science Foundation of Guangdong Province, Project No. 8451064101000320 and the Doctoral Program of Ministry of Education of China, Project No. 200805611089.

## Notation

$A$  = sectional area of channels,  $m^2$   
 $A_{pi}$  = the sectional area of the inlet,  $m^2$

$A_T$  = influence coefficient of turn angle of channels  
 $C_T$  = influence coefficient of depth-width ratio of channels  
 $D_H$  = hydraulic nominal diameter, m  
 $E$  = channel thickness, m  
 $H_{in}$  = length of  $GC_{N+1}$ , m  
 $h_{jc}$  = contraction loss  
 $h_{jd}$  = diffuser loss  
 $h_{je}$  = enlargement loss  
 $h_{ji}$  = turning loss in the inlet manifold  
 $h_{jo}$  = turning loss in the outlet manifold  
 $h_{jm}$  = mixing loss  
 $i, j$  = integer variable  
 $k$  = Slope coefficient of line segment  
 $L$  = channel length/segment length, m  
 $L_m$  = length of  $C_1C_{N+1}$ , m  
 $\max$  = maximum value  
 $\min$  = minimum value  
 $N$  = microchannel number  
 $\Delta P$  = pressure drop, Pa  
 $\Delta P_{ic}(j)$  = frictional pressure drop in the  $j$ th microchannel, Pa  
 $\Delta P_{ii}(j)$  = frictional pressure drop in the  $j$ th approximate rectangular channel of the inlet manifold, Pa  
 $\Delta P_{io}(j)$  = frictional pressure drop in the  $j$ th approximate rectangular channel of the outlet manifold, Pa  
 $\Delta P_{ji}(j)$  = singular pressure drop of entry end of the  $j$ th flow loop, Pa  
 $\Delta P_{jo}(j-1)$  = singular pressure drop of exit end of the  $j$ th flow loop, Pa  
 $Q_T$  = total feed flow rate,  $m^3 s^{-1}$   
 $Q(j)$  = flow rate of the  $j$ th channels,  $m^3 s^{-1}$   
 $R_f$  = Frictional resistance,  $Pa s m^{-1}$   
 $R_{ic}(j)$  = frictional resistance in the  $j$ th microchannel,  $Pa s m^{-1}$   
 $R_{ii}(j)$  = frictional resistance in the  $j$ th approximate rectangular channel of the inlet manifold,  $Pa s m^{-1}$   
 $R_{io}(j)$  = frictional resistance in the  $j$ th approximate rectangular channel of the outlet manifold,  $Pa s m^{-1}$   
 $R_{ji}(j)$  = singular resistance of entry end of the  $j$ th flow loop,  $Pa s m^{-2}$   
 $R_{jo}(j-1)$  = singular resistance of entry end of exit end of the  $j$ th flow loop,  $Pa s m^{-2}$   
 $R_{pi}$  = Radius of the inlet, m  
 $U$  = fluid velocity,  $m s^{-1}$   
 $U_{in}$  = entrance velocity of microchannel array model,  $m s^{-1}$   
 $U_m$  = average velocity of microchannels,  $m s^{-1}$   
 $U_T$  = Entrance velocity of the simplified resistance network model,  $m s^{-1}$   
 $W$  = channel width, m  
 $X$  = horizontal coordinate, m  
 $X_{pi}$  = horizontal coordinate of the inlet, m  
 $Y$  = vertical coordinate, m  
 $Y_{pi}$  = vertical coordinate of the inlet, m

## Subscripts

c = microchannel  
in, i = inlet manifold  
out, o = outlet manifold

## Greek letters

$\lambda_{NC}$  = non-circular coefficient  
 $\mu$  = viscosity (Pa s)  
 $\beta$  = pressure drop proportion  
 $\rho$  = fluid density ( $kg m^{-3}$ )  
 $\delta$  = turn angle of channels ( $^\circ$ )  
 $\sigma_U\%$  = estimating parameters of velocity distributions  
 $\zeta_{jc}$  = contraction loss coefficient  
 $\zeta_{jd}$  = diffuser loss coefficient  
 $\zeta_{je}$  = enlargement loss coefficient  
 $\zeta_{ji}$  = turning loss coefficient in the inlet manifold  
 $\zeta_{jo}$  = turning loss coefficient in the outlet manifold  
 $\zeta_{jm}$  = mixing loss coefficient  
 $\zeta_M$  = a factor determined by related formula  
 $\zeta_T$  = turning loss coefficients in the acute angle corner pipe with rectangular cross-section

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## Appendix A

### Calculation method of the dimensions of approximate rectangular channels

Take the inlet manifold for sample to present the calculation method of the dimensions of approximate rectangular channels with given parameters, such as  $X_{pi}$ ,  $Y_{pi}$ ,  $R_{pi}$ ,  $L_m$ , and  $H_{in}$ . The calculation method of the outlet manifold is similar to that of the inlet manifold. Some auxiliary lines are drawn to work out the length and width of each approximate rectangular channel, as shown in Figure A1. First connect G and  $C_1$  to  $P_i$ , respectively. Draw  $GG_1$  parallel to  $C_1C_{N+1}$  through G. Then draw  $P_iG_3$  perpendicular to  $C_1C_{N+1}$  through  $P_i$ , intersecting to  $GG_1$  at  $G_1$  and  $C_1C_{N+1}$  at  $G_3$ .

So, the linear equation of SG can be represented as follows

$$y = k_{SG}(x - L_m) + H_{in} \quad (A1)$$

Here,  $k_{SG}$  is the slope coefficient of SG, which is defined as below

$$k_{SG} = \tan(108^\circ - \angle T_1GG_1) \quad (A2)$$

where

$$\angle T_1GG_1 = \angle T_1GP_i + \angle P_iGG_1 \quad (A3)$$

$$\angle T_1GP_i = \arcsin \frac{L_{T_1P_i}}{L_{GP_i}} = \arcsin \frac{R_{pi}}{L_{GP_i}} \quad (A4)$$

$$\angle P_iGG_1 = \arcsin \frac{L_{P_iG_1}}{L_{GP_i}} = \arcsin \frac{Y_{pi}H_{in}}{L_{GP_i}} \quad (A5)$$

The length of  $GP_i$  is obtained as follows

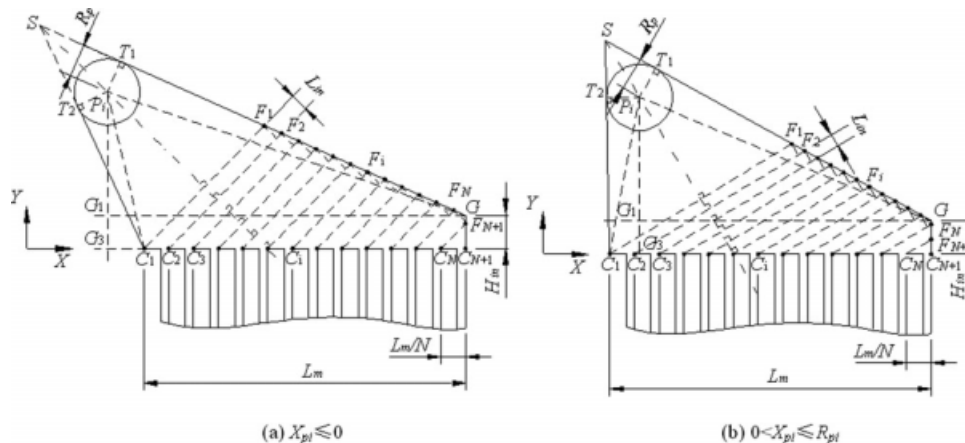


Figure A1. Auxiliary lines for calculating the lengths and widths of approximate rectangular channels.



$$L_{GP_i} = \sqrt{(X_{pi} - L_m)^2 + (Y_{pi} - H_{in})^2} \quad (A6)$$

Similarly, the linear equation of SC<sub>1</sub> can be defined as

$$y = k_{SC_1}x \quad (A7)$$

Here,  $k_{SC_1}$  represents the slope coefficient of SC<sub>1</sub>.

When the center of the inlet P<sub>i</sub> locates at the left of the axis Y or on the axis Y, i.e.,  $X_{pi} \leq 0$  as shown in Figure A1a, the magnitude of  $\angle T_2C_1G_3$  and  $k_{SC_1}$  are respectively defined as

$$\angle T_2C_1G_3 = \angle P_iC_1G_3 - \angle T_2C_1P_i \quad (A8)$$

$$k_{SC_1} = \text{tg}(180^\circ - \angle T_2C_1G_3) \quad (A9)$$

The magnitude of  $\angle T_2C_1G_3$  and  $k_{SC_1}$  are obtained as follows while  $0 < X_{pi} \leq R_{pi}$ , as presented in Figure A1b.

$$\angle T_2C_1G_3 = \angle P_iC_1G_3 + \angle T_2C_1P_i \quad (A10)$$

$$k_{SC_1} = \text{tg} \angle T_2C_1G_3 \quad (A11)$$

where

$$\angle T_2C_1P_i = \arcsin \frac{L_{T_2P_i}}{L_{C_1P_i}} = \arcsin \frac{R_{pi}}{L_{C_1P_i}} \quad (A12)$$

$$\angle P_iC_1G_3 = \arcsin \frac{L_{G_3P_i}}{L_{C_1P_i}} = \arcsin \frac{Y_{pi}}{L_{C_1P_i}} \quad (A13)$$

The length of C<sub>1</sub>P<sub>i</sub> is defined as follows

$$L_{C_1P_i} = \sqrt{X_{pi}^2 + Y_{pi}^2} \quad (A14)$$

The coordinate (X<sub>s</sub>, Y<sub>s</sub>) of S can be worked out by the combination of Eqs. A1 and A7.

$$X_s = \frac{k_{SG} \cdot L - H_{in}}{k_{SG} - k_{SC_1}} \quad (A15)$$

$$Y_s = \frac{(k_{SG} \cdot L - H_{in})k_{SC_1}}{k_{SG} - k_{SC_1}} \quad (A16)$$

The slope coefficient of SP<sub>i</sub> can be defined as

$$k_{SP_i} = \frac{Y_{pi} - Y_s}{X_{pi} - X_s} \quad (A17)$$

Then the slope coefficient of C<sub>i</sub>F<sub>i</sub> ( $i = 1, 2, \dots, N$ ) is obtained

$$k_{C_iF_i} = -\frac{1}{k_{SP_i}} \quad (i = 1, 2, \dots, N) \quad (A18)$$

The magnitude of  $\angle F_iC_iC_{N+1}$  is defined as

$$\angle F_iC_iC_{N+1} = \arctan k_{C_iF_i} \quad (i = 1, 2, \dots, N) \quad (A19)$$

It is assumed that microchannels are uniformly distributed, so the sum of each microchannel width and interval is  $L_m/N$ . So, the length of each approximate rectangular channel can be worked out.

$$L_{in} = L_m/N \cdot \sin \angle F_iC_iC_{N+1} \quad (A20)$$

The equation of C<sub>i</sub>F<sub>i</sub> is defined as

$$y = k_{C_iF_i}(x - X_{C_i}) \quad (i = 1, 2, \dots, N) \quad (A21)$$

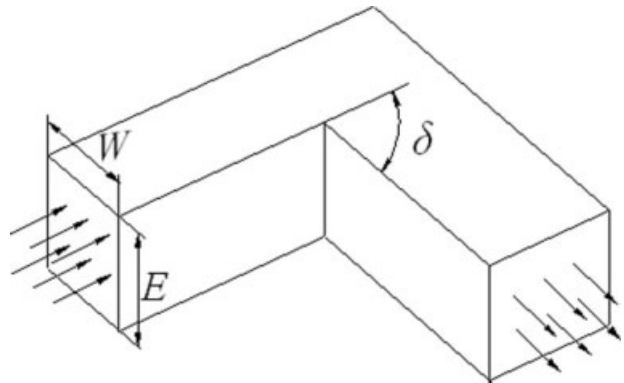
where

$$X_{C_i} = (i - 1) \frac{L_m}{N} \quad (i = 1, 2, \dots, N + 1) \quad (A22)$$

The coordinate of F<sub>i</sub> ( $i = 1, 2, \dots, N$ ) are depended on the location of the inlet and on the length  $H_{in}$ . When points F<sub>i</sub> ( $i = 1, 2, \dots, N$ ) are on the line SG, combine the Eq. A1 with Eq. A21 to solve the coordinate  $X_{F_i}, Y_{F_i}$  of F<sub>i</sub> ( $i = 1, 2, \dots, N$ ) as follows

$$X_{F_i} = \frac{k_{SG} \cdot L_m - k_{C_iF_i} \cdot (i - 1) \frac{L_m}{N} - H_{in}}{k_{SG} - k_{C_iF_i}} \quad (A23)$$

$$Y_{F_i} = \frac{-k_{SG}k_{C_iF_i}(i - 1) \frac{L_m}{N} + k_{SG}k_{C_iF_i}L_m - k_{C_iF_i}H_{in}}{k_{SG} - k_{C_iF_i}} \quad (A24)$$



**Figure B1. The schematic diagram of fluid flows in the acute angle corner pipe with rectangular cross-section.**

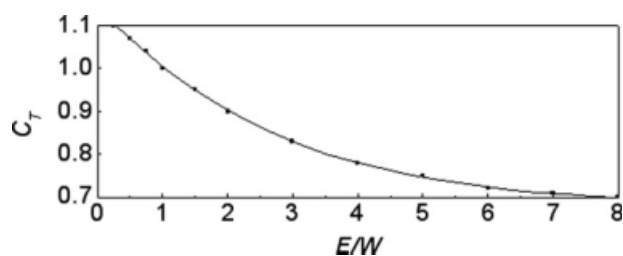


Figure B2. The curve diagram of  $C_T = f(E/W)$ .

If the value of  $X_{F_i}$  obtained by Eq. A23 is larger than that of  $L_m$ , it represents that point  $F_i$  ( $i = 1, 2, \dots, N$ ) are on the line  $GC_{N+1}$ , and the coordinate  $(X_{F_i}, Y_{F_i})$  of  $F_i$  ( $i = 1, 2, \dots, N$ ) are as follows

$$X_{F_i} = L_m \quad (\text{A25})$$

$$Y_{F_i} = k_{C_i F_i} (L_m - X_{C_i}) \quad (\text{A26})$$

Since  $Y_{C_i}$  ( $i = 1, 2, \dots, N$ ) is equal to zero, the widths  $C_i F_i$  ( $i = 1, 2, \dots, N$ ) of approximate rectangular channels can be worked out as follows

$$L_{C_i F_i} = \sqrt{(X_{C_i} - X_{F_i})^2 + Y_{F_i}^2} \quad (i = 1, 2, \dots, N) \quad (\text{A27})$$

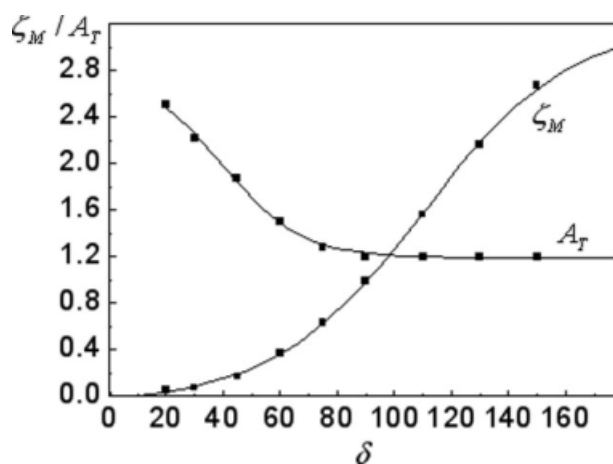


Figure B3. The curve diagram of  $\zeta_M = f(\delta)$  and  $A_T = f(\delta)$ .

## Appendix B

### *The turning loss coefficients in the acute angle corner pipe with rectangular cross-section on the large scale*

Figure B1 shows the schematic diagram of fluid flows in the acute angle corner pipe with rectangular cross-section. The turning loss is related to the depth-width ratio  $E/W$  and the turn angle  $\delta$ . In the formula,  $C_T$  relates to  $E/W$  and can be determined by Figure B2.  $A_T$  and  $\zeta_M$  relate to  $\delta$  and can be determined by Figure B3.

*Manuscript received July 6, 2008, revision received Oct. 14, 2008, and final revision received Jan. 2, 2009.*